MINISTÉRIO DA AERONÁUTICA DEPARTAMENTO DE PESQUISAS E DESENVOLVIMENTO CENTRO TÉCNICO AEROESPACIAL

TYPE CERTIFICATE DATA SHEET № EM-8012-02

Type Certificate Holder:

TURBOMECA S.A. Bordes 64320 Bizanos FRANCE

EM-8012-02

Sheet 01

TURBOMECA

ARRIEL 1A, 1A2, 1B, 1C, 1C1, 1C2, 1D, 1D1, 1E, 1E2, 1S, 1S1

December 98

Engines of models described herein conforming with this data sheet, which is part of Type Certificate No. 8012, meet the minimum standards for use in certificated aircraft in accordance with pertinent aircraft data sheets and applicable portions of the Brazilian Aeronautical Regulations provided they are installed, operated, and maintained as prescribed by the approved manufacturer's manuals and other instructions.

MODEL ARRIEL 1A, 1A2, 1B, 1C, 1C1, 1C2, 1D, 1D1, 1E, 1E2, 1S, 1S1

TYPE Twin spool (free turbine) turboshaft engine, single-stage axial compressor, single-stage centrifugal compressor, annular

combustion chamber, two-stage gas generator turbine, single-stage power turbine (free turbine), reduction gearbox with

nominal output at 6 000 rpm.

RATINGS		ARRIEL 1A	ARRIEL 1A2	ARRIEL 1B	ARRIEL 1C
	shp (kW) / See Note 8 2-1/2 minute OEI	651 (486)	670 (500)	#	700 (522)

		()	**	, , , ()
30-minute OEI	625 (466)	657 (490)	#	686 (512)
Continuous OEI	#	#	#	#
Takeoff	625 (466)	630 (470)	641 (478)	659 (492)
Maximum continuous	576 (430)	579 (432)	590 (440)	586 (437)

		ARRIEL 1C1	ARRIEL 1C2	ARRIEL 1D	ARRIEL 1D1
	shp (kW) / See Note 8 2-1/2 minute OEI	721 (538)	763 (569)	#	#
	30-minute OEI	705 (526)	738 (550)	#	#
	Continuous OEI	#	#	#	#
	Takeoff	705 (526)	738 (550)	684 (510)	712 (531)
	Maximum continuous	586 (437)	632 (471)	603 (450)	625 (466)
		ARRIEL 1E	ARRIEL 1E2	ARRIEL 1S	ARRIEL 1S1
	shp (kW) / See Note 8 2-1/2 minute OEI	626 (467)	708 (528)	751 (560)	802 (598)
	30-minute OEI	626 (467)	708 (528)	730 (544)	788 (588)
	Continuous OEI	626 (467)	708 (528)	730 (544)	788 (588)
	Takeoff	626 (467)	708 (528)	701 (523)	725 (541)
	Maximum continuous	626 (467)	692 (516)	701 (523)	725 (541)
	# Does not apply				
FUEL	See Note 15		For all	models	
FUEL CONTROL	Turbomeca		For all	models	
OIL	See Note 14		For all	models	
TEMPERATURE LIMITS	See Note 2		For all	models	
PRESSURE LIMITS	See Note 4		For all	models	
		ARRIEL 1A	ARRIEL 1A2	ARRIEL 1B	ARRIEL 1C
PRINCIPAL DIMENSIONS	Length, in.	44.2	44.2	47.7	45.9
	Width, in.	16.2	16.2	17.3	16.2
	Height, in.	23.5	24.0	23.5	24.0

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		ARRIEL 1C1	ARRIEL 1C2	ARRIEL 1D	ARRIEL 1D1
	Length, in.	45.9	45.9	49.4	47.3
	Width, in.	18.3	18.3	19.0	18.3
	Height, in.	24.0	24.0	24.0	24.1
		ARRIEL 1E	ARRIEL 1E2	ARRIEL 1S	ARRIEL 1S1
	Length, in.	46.7	46.7	60.3	60.3
	Width, in.	19.5	19.5	19.3	19.3
	Height, in.	27.4	27.4	30.9	30.9
CENTER OF GRAVITY	Refer to Installation Manual		For all	models	
		ARRIEL 1A	ARRIEL 1A2	ARRIEL 1B	ARRIEL 1C
WEIGHT	Weight / Dry / Maximun / Pounds Refer to Engine Manual for definition of dry weight:	245	257	253	257
		ARRIEL 1E	ARRIEL 1E2	ARRIEL 1S	ARRIEL 1S1
	Weight / Dry / Maximun / Pounds				
	Refer to Engine Manual for definition of dry weight:	276	276	280	286
		ARRIEL 1C1	ARRIEL 1C2	ARRIEL 1D	ARRIEL 1D1
	Weight / Dry / Maximun / Pounds Refer to Engine Manual for definition of dry weight:	262	262	272	269
DRIVE SHAFT TYPE	Refer to Installation Manual		For all	models	

IGNITION Low tension, high energy system, including:

For all models

2 high energy generators

2 injectors2 ignitors

STARTING Starting unit with electrovalve. Drain valve. (See Note 7)

For all models

IMPORT REQUIREMENTS

Each engine imported separately and/or spare parts must be accompanied by an Airworthiness Certificate for Export and/or an Airworthiness Approval Tag, respectively, issued by DIRECTION GENERALE DE L'AVIATION CIVILE (or a third country authority, in case of used engine imported from such country) attesting that the particular engine and/or parts were submitted to the governmental quality control before delivery and are in conformity with the CTA approved type design. The CTA type design corresponds to the DIRECTION GENERALE DE L'AVIATION CIVILE approved type design, as stated in applicable CTA Report V33-003x-0, where "x" corresponds to each model.

CERTIFICATION BASIS

RBHA 21.29 and 33 (Brazilian Requirements for Aeronautical Certification), which endorses the FAR Sections 21.29 and FAR 33 effective February 1, 1965, and Amendments 33-1 through 33-5.

Aplication Issued TC

ARRIEL 1A 24 May 1979 17 March 1980
ARRIEL 1A2 28 Sep. 1983 07 March 1985
ARRIEL 1B 13 Feb. 1980 17 March 1980
ARRIEL 1C 28 Sep. 1983 07 March 1985
ARRIEL 1C1 14 April 1987 10 Feb. 1993
ARRIEL 1C2 11 Oct. 1991 10 Feb. 1993
ARRIEL 1D1 14 April 1987 10 Feb. 1993
ARRIEL 1D1 21 Nov. 1989 10 Feb. 1993
ARRIEL 1E 02 Dec. 1992 10 Aug. 1993
ARRIEL 1E 20 April 1994 08 May 1995
ARRIEL 1S 01 July 1997 17 June 1998
ARRIEL 1S1 01 July 1997 17 June 1998

Arriel 1S and 1S1 are additionally certified to FAR Section 33,

Paragraphs 33.17(b), 33.67(a)(b) and 33.71(a)(b).

PRODUCTION BASIS

ARRIEL 1B, 1S, 1S1: Production Certificate Number 5SW. Produced by Turbomeca Engine Corporation in the United States under license agreement from Turbomeca S.A., France.

ARRIEL 1B, 1S and 1S1: Engine modules, and parts thereof, produced by Turbomeca S.A., France, conforming to this type certificate are fully interchangeable with ARRIEL 1B, 1S and 1S1 engine modules, and parts thereof, produced under Production Certificate Number 5SW.

Engines manufactured under Production Certificate Number 5SW shall have the suffix "TEC" added to the engine serial number and shall be included in the required identification data as specified by FAR Section 45.

NOTES

NOTE 1 PERMISSIBLE ENGINE SPEEDS / rpm
MAXIMUM GAS GENERATOR SPEED

	ARRIEL 1A	ARRIEL 1A2	ARRIEL 1B	ARRIEL 1C
2-1/2 minute OEI rating	52 700	52 900	#	53 200
30-minute OEI rating	52 000	52 250	#	52 300
Continuous OEI	#	#	#	#
Takeoff rating	52 000			51 800
Maximum continuous rating	50 750			50 500
Transient (5 second limit)	54 650			
Transient (20 second limit)	#	#	#	#

ARRIEL 1C1	ARRIEL 1C2	ARRIEL 1D	ARRIEL 1D1
53 200	53 560	#	#
52 060	52 840	#	#
#	#	#	#
52 060	52 840	52 422	52 330
50 250	50 870	50 764	50 760
54 650	55 685	54 650	55 685
#	55 685 ⁽¹⁾	#	#
ARRIEL 1E	ARRIEL 1E2	ARRIEL 1S	ARRIEL 1S1
52 888	53 509	53 517	
52 629	52 835	52 631	52 257
53 257		52 631	53 257
52 111	52 835	52 110	
51 800	51 955	52 110	
#	#	54 900	
55 685		54 900 ⁽¹⁾	
	53 200 52 060 # 52 060 50 250 54 650 # ARRIEL 1E 52 888 52 629 53 257 52 111 51 800 #	53 200 53 560 52 060 52 840 # # 52 060 52 840 50 250 50 870 54 650 55 685 # 55 685(1) ARRIEL 1E ARRIEL 1E2 52 888 53 509 52 629 52 835 53 257 52 111 52 835 51 800 51 955 # #	53 200 53 560 # 52 060 52 840 # # # # 52 060 52 840 52 422 50 250 50 870 50 764 54 650 55 685 54 650 # 55 685(1) # ARRIEL 1E ARRIEL 1E2 ARRIEL 1S 52 888 53 509 53 517 52 629 52 835 52 631 53 257 52 631 53 257 52 631 52 111 52 835 52 110 51 800 51 955 52 110 # # 54 900

For variation of these limits with outside air temperature (OAT), refer to Operation Manual or Installation Manual. For required action if limits are exceeded, refer to Operation Manual or Maintenance Manual.

100% = 51 800 rpm: Arriel 1A/1A2/1B/1C/1C1/1C2/1D/1D1/1E/1E2

100% = 52 110 rpm: Arriel 1S/1S1

(1) For one engine inoperative (OEI)

PERMISSIBLE ENGINE SPEEDS (continued) / rpm MAXIMUM POWER SHAFT SPEED POWER SHAFT SPEEDS

	ARRIEL 1A	ARRIEL 1A2	ARRIEL 1B	ARRIEL 1C
Maximum stabilized	6 780			6 480
Maximum transient (5 sec)	7 200			
Minimum transient (5 sec)	5 140			
	ARRIEL 1C1	ARRIEL 1C2	ARRIEL 1D	ARRIEL 1D1
Maximum stabilized	6 480			
Maximum transient (5 sec)	7 200			
Minimum transient (5 sec)	5 140			
	ARRIEL 1E	ARRIEL 1E2	ARRIEL 1S	ARRIEL 1S1
Maximum stabilized	6 480		6 542	
Maximum transient (5 sec)	7 200			
Minimum transient (5 sec)	5 140			

If limits are exceeded, refer to Operation Manual or Maintenance Manual

100% = 5 976 rpm: Arriel 1A/1A2/1B/1C/1C1/1C2/1D/1D1

100% = 6 057 rpm: Arriel 1S 100% = 6 409 rpm: Arriel 1S1 100% = 6 000 rpm: Arriel 1E, 1E2

See Operation Manual

- - Same as preceeding

NOTE 2 MAXIMUM PERMISSIBLE TEMPERATURE

A. EXHAUST GAS (t4°c)

Measured with 3 thermocouples on gas generator turbine diffuser

	ARRIEL 1A	ARRIEL 1A2	ARRIEL 1B	ARRIEL 1C
2-1/2 minute OEI rating	840		#	860
30-minute OEI rating	810		#	835
Takeoff rating	810			835
Maximum continuous rating	775			785
Starting	840			860
Transient (20 second limit)	#	#	#	#
	ARRIEL 1C1	ARRIEL 1C2	ARRIEL 1D	ARRIEL 1D1
2-1/2 minute OEI rating	865	885		
30-minute OEI rating	845			
Takeoff rating	845			
Maximum continuous rating	775		795	
Starting	865			
Transient (20 second limit)	#	920	#	#
	ARRIEL 1E	ARRIEL 1E2	ARRIEL 1S	ARRIEL 1S1
2-1/2 minute OEI rating	885			
30-minute OEI rating	845			868
Continuous OEI	845			868
Takeoff rating	845			
Maximum continuous rating	845 ⁽¹⁾			845
Starting	$865^{(2)}$		865	
Transient (20 second limit)	920 ⁽³⁾			

If limits are exceeded, refer to Operation Manual or Maintenance Manual for required action.

- (1) For two engine operation or one engine inoperative.
- (2) For one engine inoperative.
- (3) 5 second limit.

B. OIL / DEGREES CENTIGRADE / MEASURED AT ENGINE INLET ARRIEL 1A / 1A2 / 1B

Maximum operating temperature: 110

Minimum for starting: Between -55 and -40, according to oil and fuel specifications. Refer to Operation Manual. Minimum for power application: Between -10 and 0, according to oil specifications. Refer to Operation Manual.

MAXIMUM PERMISSIBLE TEMPERATURE (continued)

A. EXHAUST GAS (t4°c)

Measured with 3 thermocouples on gas generator turbine diffuser

ARRIEL 1C / 1C1 / 1C2 / 1D / 1D1

Maximum operating temperature: 115

Minimum for starting: Between -55 and -40, according to oil and fuel specifications. Refer to Operation Manual. Minimum for power application: Between -10 and 0, according to oil specifications. Refer to Operation Manual.

ARRIEL 1S / 1S1 / 1E / 1E2

Maximum operating temperature: Refer to Installation Manual

Minimum for starting: Refer to Installation Manual

Minimum for power application: Refer to Installation Manual

B. FUEL / DEGREES CENTRIGRADE / MEASURED AT ENGINE INLET ALL MODELS

Maximum operating temperature: 50

Minimum for starting: Between -55 and -40, according to oil and fuel specifications. Refer to Operation Manual of Installation Manual

- # Does Not Apply
- - Same as Preceeding

NOTE 3 POWER TURBINE UNIT TORQUE LIMITS / PERCENT

	ARRIEL 1A	ARRIEL 1A2	ARRIEL 1B	ARRIEL 1C
Maximum stabilized	109.2			
Maximum, 2-1/2 minutes OEI	109.2			115.4
Transient, 20 seconds	115.4			129.7
	ARRIEL 1C1	ARRIEL 1C2	ARRIEL 1D	ARRIEL 1D1
Maximum stabilized	111.1		109.2	
Maximum, 2-1/2 minutes OEI	115.4		109.2	
Transient, 20 seconds	129.7		120.0	
	ARRIEL 1E	ARRIEL 1E2	ARRIEL 1S	ARRIEL 1S1
Maximum stabilized	114.0		104.6	103.7
Maximum, 2-1/2 minutes OEI	125.0		115.0	126.7
Transient, 20 seconds	140.3		155.0	148.3

FOR 1A / 1A2 / 1B / 1C / 1C1 / 1C2 / 1D / 1D1

100 percent corresponds to 76 mdaN

FOR 1S:

100 percent corresponds to 76.4 mdaN

FOR 1S1:

100 percent corresponds to 89.16 mdaN

FOR 1E / 1E2:

100 percent corresponds to 70.25 mdaN

NOTE 4 FUEL AND OIL PRESSURE LIMITS (psig)

- A. Fuel / Refer to Operation Manual or Installation Manual
- B. Oil / Measured at engine pump outlet, after filter

ARRIEL 1A / 1A2 / 1B

Maximum 130

If limit is exceeded, refer to Operation Manual or Maintenance Manual

Minimum: At Ng between 70 percent and 85 percent: 27.6

at Ng more than 85 percent: 40.6

ARRIEL 1C / 1C1 / 1C2 / 1D / 1D1 / 1S / 1S1 / 1E / 1E2

Maximum: 72.5

If limit is exceeded, refer to Operation Manual or Maintenance Manual

Minimum: At Ng between 70 percent and 85 percent: 18.9

At Ng more than 85 percent: 26.1

NOTE 5 MAXIMUM PERMISSIBLE P2 AIR BLEED FROM CENTRIFUGAL COMPRESSOR PLENUM

Maximum air mass flow: 0.22 lb/sec

Power loss due to air bleed: Refer to Operation Manual or Performance Booklet

NOTE 6 AIR INTAKE REQUIREMENTS

The ARRIEL engines have not been tested to evaluate the affects of foreign object ingestion. Foreign object ingestion characteristics of airframe air inlet and engine combination are to be evaluated prior to approval of the engine installation.

The ARRIEL engines do not have anti-icing provisions and have not been tested to evaluate the effects of icing conditions. Anti-icing characteristics of airframe, air inlet, and engine combination are to be evaluated prior to approval of the engine installation.

The ARRIEL 1S and 1S1 engines meet the requirements of FAR 33.68(a)(b) when installed with Sikorsky S76A helicopter air intake Part Number 76302-07 001.

NOTE 7 ACCESSORY DRIVE PROVISIONS

ARRIEL 1A / 1A2 / 1B

	Direction of	Nominal	Max. Steady	Max. Torque at	Max. Stat.Overhung
Accessory / Drive	Rotation (1)	(rpm)	State Power (hp)	Overload (in.lb)	Moment Allowable For Accessories (in.lb)
Gas generator spool, compressor and turbine	$CC^{(2)}$	Ng=51 800 (100%)	#	#	#
Starter-generator/DC generator	C	Ng x $.2147 = 11126$	10	370.0	222.0
Tachometer transmitter/gas generator	CC	$Ng \times .1621 = 8401$	0.2	4.5	8.85
Oil pump unit/internal	C	$Ng \times .0910 = 4716$	2.0	122.0	#
Fuel control unit/gas generator	CC	$Ng \times .0910 = 4716$	1.0	42.0	#
Free turbine (power turbine) spool	$C^{(2)}$	NgL = 41586	#	#	#

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Alternator/AC current generator	CC	NtL x .29293=12 182	18.0	300.0	133.0
Tachometer transmitter/free turbine	CC	NtL x $.10138 = 4216$	0.2	4.5	8.85
Fuel control unit/internal/free turbine	C	NtL x $.10138 = 4216$	#	14 600	354.00
Main output shaft drive	$C^{(2)}$	NtL x. 14428 = 6 000	682	1 150	177.00
ARRIEL 1C / 1C1 / 1C2 / 1D1 / 1D1					
	Direction of	Nominal	Max. Steady	Max. Torque at	Max. Stat.Overhur
Accessory / Drive	Rotation (1)	(rpm)	State Power (hp)	Overload (in.lb)	Moment Allowab For Accessories (in.lb)
Gas generator spool, compressor and turbine	$CC^{(2)}$	Ng=51 800 (100%)	#	#	#
Starter-generator/DC generator	C	$Ng \times .2147 = 11 \cdot 126$	10	370.0	222.0
Tachometer transmitter/gas generator	CC	$Ng \times .1621 = 8401$	0.2	4.5	8.85
Oil pump unit/internal	C	$Ng \times .0910 = 4716$	2.0	122.0	#
Fuel control unit/gas generator	CC	$Ng \times .0910 = 4716$	1.0	42.0	#
Free turbine (power turbine) spool	$C^{(2)}$	NgL = 41586	#	#	#
Alternator/AC current generator	$CC^{(3)}$	NtL x .29293=12 182	18.0	300.0	133.0
Tachometer transmitter/free turbine	CC	$NtL \times .10138 = 4216$	0.2	4.5	8.85
Fuel control unit/internal/free turbine	C	$NtL \times .10138 = 4216$	#	#	#
Main output shaft drive	$C^{(2)}$	NtL x. 14428 = 6 000	682	14 600	354.00
ARRIEL 1S / 1S1 / 1E / 1E2					
Accessory / Drive	Direction of Rotation (1)	Nominal (rpm)	Max. Steady State Power (hp)	Max. Torque at Overload (in.lb)	Max. Stat.Overhu Moment Allowab For Accessories (in.lb)
Gas generator spool, compressor and turbine	CC ⁽²⁾	Ng=51 800 (100%)	#	#	#
Starter-generator/DC generator	C	Ng x $.2147 = 11188$	10	442.0	222.0

Tachometer transmitter/gas generator	C	$Ng \times .2147 = 11 188$	0.2	4.5	8.85
Oil pump unit/internal	C	$Ng \times .0910 = 4742$	2.0	122.0	#
Fuel control unit/gas generator	CC	$Ng \times .0910 = 4742$	1.0	42.6	#
Free turbine (power turbine) spool	$C^{(2)}$	$NgL = 41 981^{(4)}$	#	#	#
Alternator/AC current generator	$CC^{(3)}$	NtL x .29293=12 297	6.7	88.0	133.0
Tachometer transmitter/free turbine	CC	$NtL \times .29293 = 12297$	0.2	4.5	8.85
Fuel control unit/internal/free turbine	C	$NtL \times .10138 = 4256$	#	#	#
Main output shaft drive	$C^{(2)}$	NtL x. 14428 = 6 057	682	10 480	#

- (1) REFERENCE FACING ENGINE ACCESSORY PAD
- (2) REFERENCE AFT LOOKING FORWARD
- (3) NOT INCLUDED WITH BASIC ENGINE
- (4) CORRESPONDS TO 100% FOR 1S AND 94.5% FOR 1S1
- C CLOCKWISE
- CC COUNTER CLOCKWISE
- # Does not apply

NOTE 8 ENGINE RATINGS

Engine ratings are based on calibrated test rig with performance under the following conditions:

Static, sea level standard conditions (15°C, 29.92" Hg)

No airbleed, no accessory power extraction

6 000 rpm output shaft drive speed

Heating valve of fuel = 18 556 BTU/lb

The ratings given above are minimum final test performance of production and overhaul engines in accordance with engine acceptance test specification No. 0.292.00.940.0.

Use the exhaust pip specified below with calibrated test bed air intake No.6.202.81.719.0:

ARRIEL 1B: Exhaust pipe No. 0.292.80.721.0 or No. 0.292.80.738.0

ARRIEL 1A/1A2/1C: Exhaust pipe No. 0.292.80.753.0
ARRIEL 1C1 / 1C2 / 1D1: Exhaust pipe No. 0.292.80.818.0
ARRIEL 1D: Exhaust pipe No. 0.292.80.721.0
ARRIEL 1S/1S1/1E/1E2: Exhaust pipe No. 0.292.80.885.0

NOTE 9 FUEL SUPPLY REQUIREMENTS

Fuel supplied to the engine inlet must be filtered to 18 microns absolute efficiency per MIL-F-5504 except for ARRIEL 1S and 1S1 which has a fuel filter supplied with the engine.

Fuel icing inhibitor additive is required when operating in ambient temperatures below 0°C (32°F), except for ARRIEL 1S, 1S1, 1E, and 1E2 where fuel icing inhibitor additive is required when operating in ambient temperatures below –10°C (14°F).

NOTE 10 OIL SYSTEM

Refer to Operation Manual or Installation Manual.

NOTE 11 ENGINE MONITORING TRANSMITTERS

Refer to Operation Manual or Installation Manual.

NOTE 12 ELECTRICAL EQUIPMENT

Refer to Operation Manual or Installation Manual.

NOTE 13 ENGINE FIRE DETECTOR

Six fire detectors are provided on the engine (except for ARRIEL 1S and 1S1 which has no fire detectors installed on the engine and ARRIEL 1E, 1E2 fitted with 1 fire detector.

- **NOTE 14** Refer to Operation Manual or Installation Manual for approved oil specifications.
- **NOTE 15** Refer to Operation Manual or Installation Manual for approved fuel and additive specification.
- **NOTE 16** Life-limited components are listed in DGAC-approved Chapter 5 of the engine Maintenance Manual.

NOTE 17 MANUALS REQUIRED BY FAR 33.5

	ARRIEL 1A	ARRIEL 1A2/B	ARRIEL 1C/1C1/1C2 1D/1D1	ARRIEL 1S/1S1	ARRIEL 1E/1E1
Performance Manual No.	#		X 292 BO 001	X 292 F9 900 9	X 292 G9 9009
Installation Manual No.	292 00 931		292 02 933 2	X 292 F9 001 2	X 292 G9 002
Operation Manual No.	292 00 936		292 02 935	#	#
Maintenance Manual No.	292 01 931		292 01 939	X 292 F9 300 2	X 292 G9 3002
Overhaul Manual No.	292 01 935	292 00 935-1	292 00 935-2	X 292 87 502 2 See Note 19	See Note 19

[#] Does not apply

NOTE 18 The ARRIEL 1A may be converted to ARRIEL 1A1 by incorporating Turbomeca Service Bulletin No. 71-292-0018.

NOTE 19 Overhaul of ARRIEL 1S1, 1E and 1E2 engines is not authorized unless the appropriate overhaul manual is available; otherwise, rebuilt engines utilizing new engine tolerances may be provided by the manufacturer.

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Chefe da Divisão de Homologação Aeronáutica (Chief, Divisão de Homologação Aeronáutica)

Maj.-Brig.-do-Ar REGINALDO DOS SANTOS Diretor do Centro Técnico Aeroespacial (Director, Centro Técnico Aeroespacial)

^{- -} Same as preceeding