COMANDO DA AERONÁUTICA DEPARTAMENTO DE PESQUISAS E DESENVOLVIMENTO CENTRO TÉCNICO AEROESPACIAL

TYPE CERTIFICATE DATA SHEET Nº EM-2005T07

Type Certificate Holder:

TELEDYNE CONTINENTAL MOTORS P.O. Box 90 Mobile, Alabama 36601 USA EM-2005T07

Sheet 01

CONTINENTAL

IO-360-ES

April 2005

Engines of models described herein conforming with this data sheet, which is part of Type Certificate No. 2005T07, meet the minimum standards for use in certificated aircraft in accordance with pertinent aircraft data sheets and applicable portions of the Brazilian Aeronautical Regulations provided they are installed, operated, and maintained as prescribed by the approved manufacturer's manuals and other instructions.

MODEL IO-360-ES (See Note 4)

TYPE Six cylinders horizontally opposed, four strokes, fuel injected, spark ignition, direct drive, air-cooled, wet sump engine

incorporating a top induction system, bottom exhaust.

RATINGS IO-360-ES

Max. continuous, hp (rpm) full throttle at:

Sea level pressure altitude: 210 (2 800)

Takeoff, hp (rpm) full throttle at:

Sea level pressure altitude: 210 (2 800)

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FUEL	Minimum grade aviation gasoline	IO-360-ES 100, 100LL
OIL, LUBRICATION	TCM Specification MHS # 24	-
OIL SUMP CAPACITY	Total (qts) Usable at nose up attitude (qts) Usable at nose down attitude (qts)	8.0 5.0 (26.0°) 5.0 (18.0°)
BORE &STROKE	(in.)	4.438 x 3.875
DISPLACEMENT	(cu. in.)	360
COMPRESSION		8.5 : 1
WEIGHT (dry)	(Kg/lb)	145.15 (320.00)
C.G. LOCATION (basic engine)	Behind front face of the propeller flange (in) Below crankshaft centerline (in) Beside crankshaft centerline, toward 1-3-5 side (in)	18.70 0.46 0.14
PROPELLER SHAFT	ARP-502, Type I flange 4-7/8 in. O.D. with six 1/2 in bolts holes in 4 in. diameter circle	-
FUEL INJECTION	TCM Injector	639289A95
IGNITION	TCM / Bendix S6LN-25 TCM S6LSC-25 S6LSC-25T (right side) S6LSC-25 (left side) Slick Electro model 6214 Slick Electro model 6314	two two one one two two
TIMING	°BTC	24

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SPARK PLUGS	AC SR86, HSR86 Auto Lite SH26, SH260 Champion REM38E, REM38P, RMM RHM38P Smiths Ind. RSE-23-3R, RSH-23-3R	- - 138E, -			
TEMPERATURE LIMITS	Cylinder head bayonet, thermocouple °C Oil inlet °C(°F)	C(°F) 238 (460) 116 (240)			
PRESSURE LIMITS	Fuel pressure (psig) Inlet to injection pump, min. max. Outlet to vaporr return line, max. Oil pressure limits (psig) 2-4-6 side, Normal Idle Maximum (co	- 2.0 +8.0 +3.5 30 - 60 10 100			
IMPORT REQUIREMENTS	Each engine imported separately and/or and/or an Airworthiness Approval Tagparts were submitted to the governmentype design.	g, respectively, issued by FA	AA (USA), att	esting that the particu	lar engine and/or
CERTIFICATION BASIS	The certification basis for the engine is RBHA 33, corresponding to FAR 33	Amendment 1 through 12 effective on 02 September 1988.	Model IO-360-ES	Application 1 September 2004	Issued TC 15 April 2005

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NOTES:

NOTE 1 Accessory drive or mounting provisions:

	Accessory	Direction of rotation	Speed Ratio to	Max. Torq	ue (inlb.)	Max. Overhang
		(*)	Crankshaft	Cont	Static	Moment (inlb.)
IO-360-ES	Propeller governor	С	1:1	29.0	825.0	50.0
	Tachometer (mech. opt.)	C	0.495:1	-	-	-
	Magneto	C	1.5:1	-	-	-
	Alternator	CC	2.035:1	60.0	100.0	100.0
	Fuel injection pump	CC	1:1	-	-	-
	Opt.Compr. drive (V-belt)	C	3.25:1	110.0	400.0	-
	Opt. Alt. drive (V-belt)	C	3.18:1	110.0	400.0	-
	Generator	CC	2.035 : 1	60.0	600.0	50.0

(*) "C" Clockwise viewing drive pad; "CC" Counter Clockwise.

NOTE 2 TCM Eq. Nº 6001 oil filter adapter is eligible for installation.

NOTE 3 TCM P/N 627841 (Delco-Remy P/N 1108234) starter is eligible for installation.

NOTE 4 Additional suffix to model designation on engine data plate defines accessory configuration and do not affect eligibility. Example: IO-360-ES4.

NOTE 5 Model IO-360-ES incorporates a crankshaft with one 6^{th} and one $4 \frac{1}{2}^{th}$ order dampers.

NOTE 6 Model IO-360-ES is eligible for pusher and tractor operation and are approved for installation of propellers or propeller-fan combination having inertias up to 20.-lb sec. and overhang moments up to 490 in.-lb. The maximum overhung weight and moment are 70 lb. and 8 in. respectively.

NOTE 7 Applicable FAA approved and/or accepted manuals: Maintenance & Operator's Overhaul X30594A

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