COMANDO DA AERONÁUTICA DEPARTAMENTO DE PESQUISAS E DESENVOLVIMENTO CENTRO TÉCNICO AEROESPACIAL

TYPE CERTIFICATE DATA SHEET No. EH-8808-02

Type Certificate Holder:

HARTZELL PROPELLER INC.

One Propeller Place Piqua - OH - 45356 -2634 USA EH-8808-02

Sheet 01

HARTZELL

HC-B4M

December 2001

Propellers of models described herein conforming with this data sheet, which is part of Type Certificate No.8808, meet the minimum standards for use in certificated aircraft in accordance with pertinent aircraft data sheets and applicable portions of the Brazilian Aeronautical Regulations provided they are installed, operated, and maintained as prescribed by the approved manufacturer's manuals and other instructions.

TYPE Constant Speed, Hydraulic (See Notes 3 & 4).

ENGINE SHAFT Special flange 4-1/4" bolt circle with eight 9/16" bolts.

HUBMATERIAL Alloy steel.

BLADE MATERIAL See page 2.

NUMBER OF BLADES Four.

HUBS ELIGIBLE Hub Models HC-B4MN-5 and HC-B4MP-3 (See Notes 1 and 4).

Blade Eligible (See Note 2).	Max.Continuous Power		Take off power		Diameter Limits		Approx. Max. Weight Compl. (See Notes 3 and 7)		Blade Construction
	hp	rpm	hp	rpm	m	in	Kg	lb	
N10476	1200	1700	1200	1700	2,67	105 To	07.00	102.0	A1 · 11
M10476	1200	1700	1200	1700	2,41	95 (co -10)	87,09	192.0	Aluminum alloy
					(-0 (.0 -10)			
M10585+4	900	1591	940	1591	2,79	110 ТО	77,11	170.0	Aramid Composit
to M10585-0					2,69 (+4	106 to 0)			
M10877	1173	1700	1173	1700	2,78	109.5	77,56	171.0	Aramid Composit
M10877 S	1173	1700	1173	1700	2,78	109.5	78,47	173.0	Aramid Composit

CERTIFICATION BASIS

RBHA 35 (Brazilian Requirements for Aeronautical Certification), which endorses the FAR Part 35 effective May 2, 1977, with Amendments 1 through 4.

TYPE CERTIFICATION	Aplication	Issued TC
HC-B4MP-3	19 Nov.1987	29 July 1988
HC-B4MN-5	26 July 2001	05 December 2001

PRODUCTION BASIS

Production Certificate No. 10.

IMPORT REQUIREMENTS

Each propeller imported separately and/or spare parts must be accompanied by an Airworthiness Certificate for Export and/or an Airworthiness Approval Tag, respectively, issued by FAA, attesting that the particular propeller and/or parts were submitted to the governmental quality control before delivery and are in conformity with the CTA approved type design.

NOTES

NOTE 1

<u>Hub model Designation - HC B 4 M N 5 A L</u>, where:

HC Hartzell Controllable.

B Identifies basic design.

4 Number of blades.

5

M Hartzell blade shank size.

N denotes Special Flange 4-1/4" bolt circle with eight 9/16" bolts and two dowel bushings while P denotes 4 dowel bushings on N flange details.

5 Denotes specific design features (See Note 4).

A Denotes minor change not affecting eligibility.

L Denotes left-hand rotation.

NOTE 2 Blade Model Designation - \underline{L} \underline{M} 105 85 \underline{A} \underline{N} \underline{K} +4, where:

L Denotes left hand blade.

M Denotes needle bearing installation in blade shank.

Basic diameter in inches.

Basic model or template.

A A denotes blade cuff modification.

N denotes nickel erosion shield S on aramid composite blade represents a stainless steel wire screen on blade surface.

K B or K denotes deicing boots.

4 Number of inches cut off from or added (+) to basic diameter.

NOTE 4 Feathering. – The -3 and -5 models incorporate feathering and unfeathering features.

Reversing. – The -3 and -5 models are approved for installation as reversing propellers with appropriate reversing controls.

NOTE 5 Left-hand Model.

The left-hand version of an approved model propeller is approved at the same rating and diameter as listed for right-hand model. (See Notes 1 and 2.)

NOTE 6 <u>Interchangeability</u>: Not Applicable.

NOTE 7 Accessories:

- (a) Propeller Spinner: Approved with Hartzell spinners (weight of spinners extra).
- (b) Propeller deicing: Approved with Goodrich 77-xxx or 65-xxx deicing kit when installed in accordance with manufacturer's instructions.

NOTE 8 Shank Fairings.

The basic design incorporated a shank fairing.

NOTE 9 Special Limits. (Not Applicable).

NOTE 10 Special notes.

Propeller installation must be approved as part of the aircraft Type Certificate and demonstrate compliance with the applicable aircraft airworthiness requirements.

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