



TYPE CERTIFICATE DATA SHEET Nº EM-2023T04

Type Certificate Holder:

Lycoming Engines

An Operating Division of AVCO Corporation
Williamsport, Pennsylvania 17701
USA

EM-2023T04-00

Sheet 01

LYCOMING ENGINES

IO-320-D1B

31 August 2023

Engines of models described herein conforming with this data sheet, which is part of Type Certificate No. 2023T04, meet the minimum standards for use in certificated aircraft in accordance with pertinent aircraft data sheets and applicable portions of the Brazilian Aeronautical Regulations provided they are installed, operated, and maintained as prescribed by the approved manufacturer's manuals and other instructions.

MODEL IO-320-D1B

TYPE 4HOA Direct Drive - the Lycoming Engines models of the IO-320 series are a fuel injection Direct Drive, four cylinders, horizontally opposed, air cooled engines. These engines are supplied at the factory with automotive type alternator and starter. Available as optional equipment are a propeller governor type, a vacuum pump and the hydraulic pump drives.

RATINGS IO-320-D1B

Max. continuous, kW (hp.) - rpm. full throttle at:

Sea level pressure altitude:

119 kW (160 hp.)

at 2700 engine rpm

Takeoff, kW (hp.) - rpm full throttle at:

Sea level pressure altitude:

119 kW (160 hp.)

at 2700 engine rpm

FUEL TYPE Minimum grade aviation gasoline, see latest revision of Lycoming Service Instruction No. 1070

OIL, LUBRICATION Latest revisions of Lycoming Spec.No. 301 and Service Instruction 1014
Obs.: Lubricants should conform to the specification as listed or subsequent revisions)

BORE AND STROKE, mm (in.) 130.2 x 98.4 (5.125 x 3.875)

DISPLACEMENT, cm³ (cu.in.) 5,243 (320)

COMPRESSION RATIO 8.50:1

WEIGHT DRY, Kg (lb.) 119 (263)
Standard engine dry weight less starter and generator/alternator.

WEIGHT AND C.G. LOCATION From front face of Prop. Shaft Flange, mm (in.) → 370.6 (14.59)
Off Crankshaft Center Line, mm (in.) → 25.9 (1.02) Below – 2.03 (0.08) Left

PROPELLER SHAFT FLANGS, SAE No.	AS 127 Type 2 modified
CRANKSHAFT DAMPERS AND BALACERS	None
FUEL INJECTOR	Bendix RSA-5AD1
IGNITION, DUAL	Bendix S4LN-1209, S4LN-1208 For alternate magnetos see latest issue of Lycoming Service Instruction 1443.
TIMING, °BTC	25
SPARK PLUGS	See latest revision of Lycoming Service Instruction No. 1042 for approved equipment.
OIL SUMP CAPACITY, liters (qt.)	7.6 (8)
MINIMUM USABLE OIL, liters (qt.) (30° nose up or down)	1.9 (2)
MINIMUM USABLE OIL, liters (qt.) (30° nose up, 20° nose down)	Does not apply.

IMPORT REQUIREMENTS Each engine imported separately and/or spare parts must be accompanied by a FAA Export Airworthiness Approval, Authorized Release Certificate (or a third country authority, in case of used engine imported from such country), certifying that the engine conforms to a type design approved by the ANAC, as specified in the ANAC type certificate data sheet (TCDS) No. EM-2023T04-latest revision, is in condition for safe operation and has undergone a final operational check. The original Authorized Released Certificate should be sent with the engine and a copy remains with the issuing organization.

For each engine it is required a list of exceptions (if any) in respect to the ANAC approved Type Design, listed in the FAA Authorized Release Certificate above mentioned.

CERTIFICATION BASIS Brazilian Type Certificate No. 2023T04 is based in RBAC 21 - Certificação de Produto e Artigo Aeronáuticos (Certification Procedures for Products and Articles), Section §21.29, including the following Airworthiness Requirements:

- CAR 13 effective June 15, 1956. As amended by 13-1, 13-2, 13-3 & 13-4.

<u>Model</u>	<u>Application</u>	<u>Issued TC</u>
IO-320-D1B	05 Jul. 2023	31 Aug. 2023

STATE OF DESIGN AUTHORITY REFERENCE DOCUMENT TCDS FAA 1E12; Rev. 10, 04 April 2018

PRODUCTION BASIS FAA Production Certificate no. 3

NOTES:**NOTE 1 Maximum permissible temperatures, °C (°F):**

	Temperature in °C / °F	Comments
Cylinder head	260 °C / 500 °F	Well-type thermocouple
Cylinder Base	163 °C / 325 °F	Not applicable to engine models which incorporate internal piston cooling oil jets
Oil inlet	118 °C / 245 °F	

NOTE 2 Pressure Limits:**Fuel:**

Inlet to Diaphragm Pump		Inlet to Injector	
Maximum	Minimum	Maximum	Minimum
35 p.s.i.	-2 p.s.i.	45 p.s.i.	12 p.s.i.

Boost pump outlet limits to injectors:

	Parallel Boosts		Series Boosts	
	Maximum	Minimum	Maximum	Minimum
Zero Fuel Flow	45 p.s.i.	-	35 p.s.i.	-
Maximum Fuel Flow	-	14 p.s.i.	-	14 p.s.i.

Oil:

	Maximum	Minimum
Normal Operating	Starting and Warm-up	Normal
90 p.s.i.	100 p.s.i.	60 p.s.i.
		Idling
		25 p.s.i.

NOTE 3 The following accessory provisions are incorporated:

Accessory	Rotation Facing Drive Pad	Speed Ratio to Crankshaft	Max. Torque N.m (in.lb)		Maximum Overhang Moment N.m (in.lb)
			Cont	Static	
Starter *	CC	13.556:1	-	50.8 (450)	16.9 (150)
Starter **	CC	16.556:1	-	50.8 (450)	16.9 (150)
Alternator *	C	3.250:1	6.8 (60)	13.6 (120)	19.8 (175)
Fuel Pump, plunger *	-	0.5:1	-	-	1.1 (10)
Vacuum Pump *	CC	1.300:1	7.9 (70)	50.8 (450)	2.8 (25)
Tachometer*	C	0.5:1	0.8 (7)	5.7 (50)	0.6 (5)
Hydraulic Pump *	C	1.300:1	11.3 (100)	90.4 (800)	4.5 (40)
Propeller Governor *	C	0.866:1	14.1 (125)	135.6 (1200)	4.5 (40)

"C" = Clockwise, "CC" = Counterclockwise

* Standard, ** Optional

NOTE 4 Reserved**NOTE 5** This engine incorporates provisions for absorbing propeller thrust in both tractor and pusher type installations.**NOTE 6** This engine is approved for horizontal helicopter application and operation.

NOTE 7 Reserved

NOTE 8 Starters, generators and alternators approved for use on these engines are listed in the latest revision of AVCO Lycoming Service Instruction No. 1154.

NOTE 9 Reserved

Change Record

Revision	Application Date	Changes	TC Issue/Reissued Date
Rev. 00	05 July 2023	Original Issue	31 Aug. 2023

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This TCDS is available at ANAC website:

<https://sistemas.anac.gov.br/certificacao/Produtos/EspecificacaoOrgE.asp>