MINISTÉRIO DA AERONÁUTICA DEPARTAMENTO DE PESQUISAS E DESENVOLVIMENTO CENTRO TÉCNICO AEROESPACIAL

TYPE CERTIFICATE DATA SHEET Nº EH-8802

Type Certificate Holder

McCAULEY ACCESSORY DIVISION CESSNA Aircraft Company Vandalia, Ohio, 45377 USA EH-8802-02

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McCAULEY

2A34C, B2A37C

OCTOBER , 1992

The propeller model described herein conforming with this data sheet which is part of Type Certificate n^{O} 8802 prescribes conditions and limitations under which the propeller for which the type certificate was issued meets the minimum standards for use in certificated aircraft in accordance with pertinent aircraft type certificate data sheets and airworthiness requirements of the applicable portions of the Brazilian Regulations for Aeronautical Certification – RBHA – provided they are installed, operated, and maintained as prescribed by the approved manufacturer's manuals and other approved instructions.

- TYPE
Engine shaft

Hub and Blade Material Number of blades

Hubs eligible

Constant speed; hydraulic (see NOTE 3)

Special flange 10.16 cm (4.0 inches) B.C. Aluminium Alloy

Two

2A34C221, B2A37C228

- TABLE 1 - Hub model 2A34C221

Blades eligible (See NOTE 2)	Max. Cont. kw/rpm (shp)	Takeoff kw/rpm (shp)	Diameter Limits (See NOTE 2) mm (in)	Approx. Max. (For reference kg (1	
90 DH(x)-12 to 90 DH(x)-20	156/2700 (210)	156/2700 (210)	1980 (78") to 1780 (70")	24,4 (5	53.8)

- TABLE 2 - Hub model B2A37C228

Blades eligible (See NOTE 2)	Max. Cont. kw/rpm (shp)	Takeoff kw/rpm (shp)	Diameter Limits (See NOTE 2) mm(in)	Approx. Max. Wt. (For reference only) kg (lb)
90 RD(x)-0 to	224 (2700)	224/2700	2286 (90") to	27,7 (61)
90 RD(x)-18	(300)	(300)	1829 (72")	

- CERTIFICATION BASIS Model 2A34C221/90 DHC:

FAR Part 35, effective October 14, 1980 with amendments 35-1 thru 35-5 thereto.

Model B2A37C228/90RD(x):

FAR Part 35 effective August 18, 1990 with amendments 35-1 thru 35-6 thereto.

 DATE OF APPLICATION FOR TYPE CERTIFICATE,

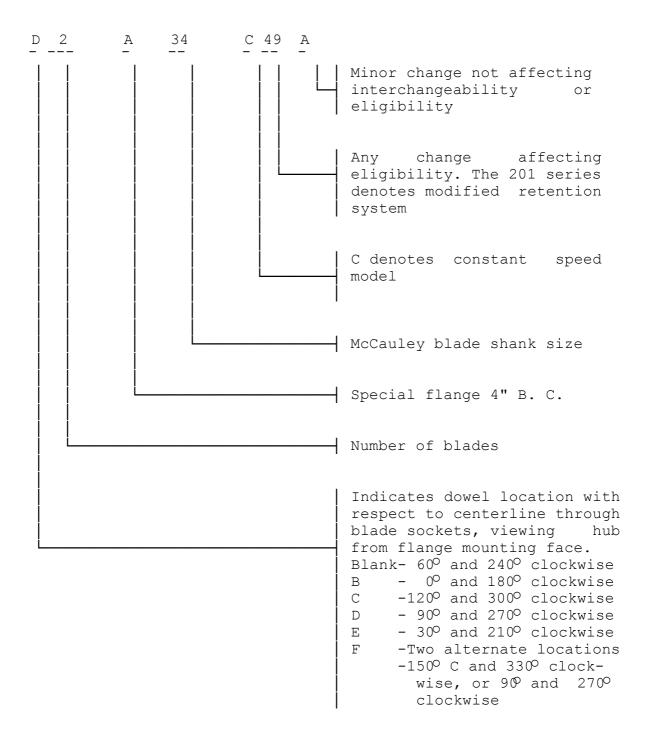
January 15, 1987

Type Certificate n° 8802 issued in Aug 30,91 Production Certificate \Re . 3 (FAA)

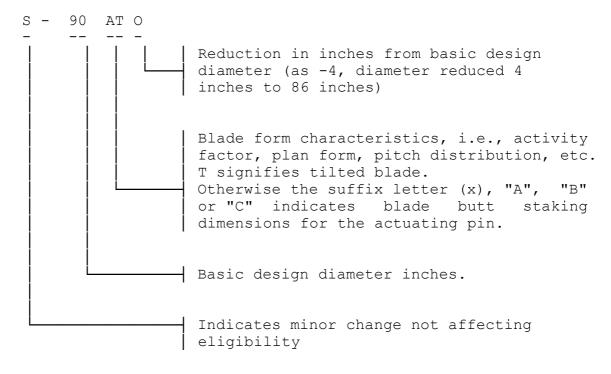
- IMPORT REQUIREMENTS

Each propeller exported separately must be accompanied by an Airworthiness Certificate for Export issued by the Federal Aviation Administration attesting that the particular propeller was submitted to governmental quality control before delivery and is in conformity with the adopted certification basis.

NOTE 1 - HUB MODEL DESIGNATION



NOTE 2 - BLADE MODEL DESIGNATION



- NOTE 3: Pitch Control - Eligible with the following governors.

Woodward Model x, 210, xxx	wt.	1,59	kg	(3.5	lb)
Hoof Model 1,000,007 series	wt.	1,59	kg	(3.5	lb)
Hoof Model 1,604,014 series	wt.	1,59	kg	(3.5	lb)
Garwin Model 34 - 828 series	wt.	1,36	kg	(3.0	lb)
Hartzell Model F-2-7 ()	wt.	2,04	kg	(4.5	lb)
McCauley Model C290D(x)/T(x)	wt.	1,27	kg	(2.8	lb)

- NOTE 4 : Not applicable.

- NOTE 5 : LEFT HAND MODELS

The left hand version of an approved model is eligible at the same rating and diameter limitations as listed for the right hand model.

- NOTE 6 : Not applicable.

- NOTE 7 : ACCESSORIES

- (a) Spinners
 - (1) approved McCauley spinners; D-3396 or D-5255 Installation for B2A37C228 model.
- NOTE 8 : Not applicable.

- NOTE 9 : TABLE OF PROPELLER - ENGINE COMBINATIONS

APPROVED VIBRATIONWISE FOR USE ON NORMAL CATEGORY SINGLE - ENGINE TRACTOR AIRCRFT.

The maximum and minimum propeller diameters that can be used from a vibration standpoint are shown below. No reduction below the minimum diameter listed is permissible, since this figure includes the diameter reduction allowable for repair purposes.

	Blade Model	Engine Model	Crankshaft Damper Configuration
2A34C221	90DHC	Continental TSIO-360 Series up to 113.8kw(210HP) 2700 rpm	Order Crankshaft
Max. Dia. mm (inches)		in. Dia. m (inches)	Placards
1879.6mm (7	'4 ") 1	828.8mm (72")	None
Hub	Blade	Engine	Crankshaft Damper

Hub Model	Blade Model	Engine Model	Crankshaft Damper Configuration	
*B2A37C228	90RD(x) - 0 to 90RD(x) - 18	Continental IO-550-E 221.0kw(300HP) 2700rpm	2-6, 1-5 and 1-4 Order Crankshaft Dampers.	· -
Max. Dia mm (inches)		. Dia. Pi (inches)	lacards	
2184.4mm (8	86") 213	3.6mm (84")	None	

^{*} This propeller/engine combination as noted is only approved when installed in the Neiva EMB-200/200A/200R/201/201A/202 - IPANEMA aircraft.

- NOTE 10 : SPECIAL NOTES

The word "eligible" as used herein does not signify approval as part of this Type Certificate, "Eligible" acessories must be approved as part of the aircraft type certificate upon compliance with the applicable aircraft airworthiness requirements.

PAULO GASTÃO SILVA - Maj Eng Chefe da Divisão de Homologação Aeronáutica Maj Brig do Ar - ADYR DA SILVA Diretor do CTA

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