

TYPE CERTIFICATE DATA SHEET Nº EA-2012T01

Type Certificate Holder:

QUEST AIRCRAFT DESIGN. LLC 1200 Turbine Drive Sandpoint, Idaho 83864 USA

EA-2012T01-00 Sheet 01

QUEST

KODIAK 100

19 January 2012

This data sheet, which is part of Type Certificate No. 2012T01, prescribes conditions and limitations under which the product, for which the Type Certificate was issued, meets the airworthiness requirements of the Brazilian Aeronautical Regulations.

I - Model Kodiak 100 (Normal Category), approved 19 January 2012.

ENGINE

1 Pratt & Whitney Canada, Inc. PT6A-34 (EM-8005)

FUEL

Jet A, Jet A-1, JP-1, JP-5, and JP-8 Fuels conforming to Pratt and

Whitney Canada Specification CPW204.

ENGINE LIMITS

Maximum takeoff power = 750 horsepower at 2,200 rpm, subject to ambient temperature and pressure conditions. See Airplane Flight Manual and Engine Type Certificate Data Sheet EM-8005

for additional limitations.

OIL

See Pratt & Whitney Canada Engine Service Bulletin Number

1001 for type of oil.

PROPELLER AND PROPELLER LIMITS

Model HC-E4N-3P(Y)/D9511FSB, Propeller Hartzell

Certificate 9107

Minimum diameter = 2.41 m (95 in) Maximum diameter = 2.44 m (96 in)

Low Pitch = $17.5^{\circ} \pm 0.5^{\circ}$ High Pitch = 86.1° ± 0.5° Reverse = $-10.0 \pm 0.5^{\circ}$

Pitch limits measured at 0.76 m (30 in) radial distance.

Stabilized ground operation is prohibited between 450 and 1050

See Propeller Type Certificate Data Sheet EH-9107 for additional

AIRSPEED LIMITS (CAS)

Maximum operating (V_{MO}) :

180 kcas (182 kias)

Type

Maneuvering (V_A) - sea level:

143 kcas (142 kias)

Flaps extended (V_{FE})

- 35° (landing):

108 kcas (108 kias)

- 20° (takeoff and approach):

120 kcas (120 kias)

- 10° (takeoff):

139 kcas (138 kias)

CG RANGE

(Straight line variation between

points.)

Aft Limits 2.05 m (80.775 in) aft of datum from 1 846 kg (4 070 lb)

to 3 291 kg (7 255 lb)

Forward Limits 1.62 m (63.896 in) aft of datum from 1 846 kg (4 070 lb) to 2 268 kg (5 000) lb then to 1.80 m (71.037 in) aft of

datum at 3 291 kg (7,255 lb)

DATUM The forward face of the firewall represents datum 0.00 m (0.00 in).

LEVELING MEANS Place a level on the seat tracks in the aft cabin next to the cargo

door forward post. See latest approved revision of "Pilots

Operating Handbook and ANAC Approved Flight Manual".

MAXIMUM WEIGHT Takeoff: 3 291 kg (7 255 lb)

Landing: 3 035 kg (6 690 lb)

Zero Fuel: 2 944 kg (6 490 lb)

Ramp: 3 314 kg (7 305 lb)

Design minimum flight weight: 1 846 kg (4 070 lb)

MINIMUM CREW 1 pilot

NO. OF SEATS Up to 10 seats total, including 2 seats located at 1.02 m (40 in) aft

of datum and up to 8 additional seats located in accordance with latest approved revision of "Pilots Operating Handbook and ANAC

Approved Flight Manual"

MAXIMUM BAGGAGE As defined in latest approved revision of "Pilots Operating

Handbook and ANAC Approved Flight Manual".

FUEL CAPACITY 1211.3 I (320 gal) total; 1192.4 I (315 gal) useable.

(Two 605.65 I (160 gal) tanks in wings at 2.12 m (83.4 in) aft of

datum).

OIL CAPACITY 8.52 I (9 qts) drainable, 12.3 I (13 qts) total.

MAXIMUM OPERATING ALTITUDE 4 267 m (14 000 ft) without ANAC approved oxygen system

installed.(max. 30 min)

7 620 m (25 000 ft) with ANAC approved oxygen system installed.

CONTROL SURFACE

MOVEMENTS

Elevator: Up $30^{\circ} \pm 1^{\circ}$ Down $22^{\circ} \pm 1^{\circ}$ Elevator trim tab: Up $15^{\circ} \pm 2^{\circ}$ Down $12^{\circ} \pm 1^{\circ}$ Rudder: Right $26^{\circ} \pm 1^{\circ}$ Left $26^{\circ} \pm 1^{\circ}$

Aileron: Up $28^{\circ} \pm 1^{\circ}$ Down $15^{\circ} \pm 1^{\circ}$ Aileron trim tab: Up $30^{\circ} \pm 2^{\circ}$ Down $30^{\circ} \pm 2^{\circ}$

Wing flaps: Down $0^{\circ} \pm 0^{\circ}$, $10^{\circ} + 1^{\circ} -2^{\circ}$, $20^{\circ} \pm 2^{\circ}$, $35^{\circ} \pm 10^{\circ}$

2°,

ADDITIONAL LIMITATIONS Minimum Operating OAT -55°C for aircraft Serial Numbers 100-

0018 and on and Serial Numbers 100-0001 thru 100-0017 that have complied with Quest KODIAK Service Bulletin SB-016 (FLAP

SYSTEM, Flap Drive Actuator Replacement).

Minimum Operating OAT -25°C for Aircraft 100-0001 thru 100-0017 that have not complied with Quest KODIAK Service Bulletin

SB-016 (FLAP SYSTEM, Flap Drive Actuator Replacement).

Airframe life limit: 1,000 flight hours for airplane serial numbers 100-0001 through 100-0036 except as described below.

5000 hours for airplane serial numbers 100-0001 through 100-0036 when the latest FAA approved revision of Service Bulletin SB-028 has been accomplished.

5000 hours for airplane serial numbers 100-0037 and on.

Kinds of operations allowed: Day, Night, Visual Flight Rules (VFR), and Instrument Flight Rules (IFR). Flight into known icing conditions allowed when the required equipment listed in Quest POH/AFM Supplement 6 "TKS Known Icing System" was installed when the airplane was produced (may have required system activation in accordance with Quest Service Notice SN-043) or installed in accordance with the latest FAA approved revision of Quest Field Service Instruction FSI-013.

For Brazilian registered aircraft S/N 100-0001 thru 100-0034 the Kodiak Service Notice SN-025 must be accomplished.

SERIAL NUMBER ELIGIBLE

100-0001 and on

IMPORT ELIGIBILITY

A Brazilian Certificate of Airworthiness may be issued on the basis of an FAA Export Certificate of Airworthiness (or a third country Export Certificate of Airworthiness, in case of used aircraft imported from such country), including the following statement:

"The aircraft covered by this certificate has been inspected, tested, and found to be in conformity with the Brazilian approved type design as defined by the Brazilian Type Certificate no. 2012T01 and in condition of safe operation".

CERTIFICATION BASIS

Brazilian Type Certificate No. 2012T01 issued on 18 January 2012 based on the RBAC 21.29 and RBAC 23, which endorses the 14 CFR Part 23 effective 01 February 1965, as amended by 23-1 through 23-55.

Special Conditions:

Special Condition 23-207-SC for high intensity radiated fields (HIRF) protection, effective January 31, 2007.

Noise requirements:

FAR 36 as amended through 36-28. Application for type certificate dated April 7, 2005.

REQUIRED EQUIPMENT

The basic required equipment, as prescribed in the applicable airworthiness regulations (see Certification Basis) must be installed in the airplane.

DATA PERTINENT TO ALL MODELS:

NOTES:

NOTE 1 Weight and balance.

A current weight and balance report with a list of equipment included in the certificated empty weight must be provided for each aircraft at the time of original airworthiness certification.

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NOTE 2 Markings and placards.

For the approved markings and placards translations contact the TC holder and/or ANAC at the following address: ggcp-gr@anac.gov.br

- NOTE 3 <u>Continuing Airworthiness</u>. The airplane must be maintained in accordance with the instructions for continued airworthiness contained in the latest FAA approved revision of "Airplane Maintenance Manual".
- NOTE 4 The differences of the Brazilian airplanes in relation to the basic FAA type design are summarized below:
 - 1. The Brazilian Airplane Flight Manual Supplement

2. Markings and placards

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