

AGÊNCIA NACIONAL DE AVIAÇÃO CIVIL - BRAZIL

BRAZILIAN AIRWORTHINESS DIRECTIVE

AD No.: 2024-05-05 Effective Date: 21 Jul. 2024

The following Brazilian Airworthiness Directive (AD), issued by the Agência Nacional de Aviação Civil (ANAC) in accordance with provisions of Chapter IV, Title III of Código Brasileiro de Aeronáutica - Law No. 7,565 dated 19 December 1986 - and Regulamento Brasileiro da Aviação Civil (RBAC) 39, applies to all aircraft registered in the Registro Aeronáutico Brasileiro. No person may operate an aircraft to which this AD applies, unless it has previously complied with the requirements established herein.

AD No. 2024-05-05 - (BOEING) / 39-1556.

APPLICABILITY:

(a) This Airworthiness Directive (AD) applies to BOEING airplane models 787-8, 787-9 e 787-10, all serial numbers.

CANCELLATION / REVISION:

Not applicable.

REASON:

The reason for this AD is the finding of potential interference in radio altimeters from wireless broadband operations in the 3,300 MHz to 3,700 MHz frequency band (5G C-Band). During landings, as a result of this interference, certain airplane systems may not properly function, resulting in longer than normal landings due to the effect on thrust reverser, speedbrake deployment, and increased idle thrust, regardless of the approach type or weather. The degraded deceleration performance could lead to a runway excursion.

Since this condition may occur in other airplanes and affects flight safety, corrective action is required. Thus, sufficient reason exists to mandate compliance with this AD in the indicated time limit.

REQUIRED ACTION:

Airplane Flight Manual (AFM) Revision

COMPLIANCE:

Required as indicated below, unless already accomplished.

(b) Airplane Flight Manual Revision

(1) For airplanes identified in paragraph (a) of this AD, that do not meet

the criteria for a "radio altimeter tolerant aircraft", as established by PORTARIA No. 14.318/SAR, de 10 de abril de 2024, or further Portaria that supersedes it, within 10 days after the effective date of this AD, revise the Limitations Section of the existing AFM to include the following information:

Radio Altimeter 5G C-Band Interference, Landing Performance

Due to the presence of 5G C-Band wireless broadband interference, the following limitations are required if dispatching or releasing to or landing on runways, in the Brazilian airspace.

Minimum Equipment List (MEL)

Dispatch or release with any of the following MEL items is prohibited:

- 32-42-02 Antiskid Control Systems
- 32-45-01 Wheel Brake Systems
- 32-45-01-01 Wheel Brake Systems, Electric Brake Actuator Systems

Landing Operations on Runways with ice, wet ice, water on top of compacted snow, dry snow, or wet snow over ice

Dispatch or release to, or takeoff or landing on, runways with ice, wet ice, water on top of compacted snow, dry snow, or wet snow over ice is prohibited.

Takeoff and Landing Performance

Operators must use the **5G C-Band Interference Landing Distance** Calculations procedure contained in the Operating Procedures Section of this AFM.

(2) For airplanes models identified in paragraph (a) of this AD, that do not meet the criteria for a "radio altimeter tolerant aircraft", as established by PORTARIA No. 14.318/SAR, de 10 de abril de 2024, or further Portaria that supersedes it, within 10 days after the effective date of this AD, revise the Operating Procedures Section of the existing AFM to include the following information:

5G C-Band Interference Landing Distance

Dispatch or release:

- No additional landing distance calculations are required for dry runway conditions.
- For wet runway conditions, use Tables 1 through 6, as applicable, to determine the
 unfactored landing distance, applying all adjustments. Multiply the resulting
 unfactored landing distance by 1.15 to obtain the minimum required landing
 distance.

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400		L	anding Dist	ance and Ad	justments (f	eet)			
	Reference Distance	Weight adjustment	Altitude adjustment	adjustment adjustment	Slope Adjustment per 1%	Temperature Adjustment per 10° C	Approach Speed Adjustment	Reverse Thrust Adjustment	
Runway condition	440,000 LB Landing Weight	Per 10,000 LB Above Below 440,000 LB	Per 1,000 ft	Head/Tail wind	Down/Up Hill	Above/Below ISA	Per 5 KTS above VREF	One Reverser	No reverse
Dry	5640	1104-90	160	-240/790	90/-80	150/-150	230	0	0
Wet (non- contaminated)	9180	170/-150	340	-470/1680	440/-340	290/-280	390	0	0
Wet(contaminated)	12180	280/-250	560	-770/2850	970/-690	480/-460	540	0	0

Table 2:

		1	Landing Dis	tance and A	djustments (f	eet)			
Dist Rumway condition 440,0 Lan	Reference Distance	Weight Altitude Wind adjustment adjustment per 10 knots	Slope Adjustment per 1%	tment Adjustment	Approach Speed Adjustment	Reverse Thrust Adjustment			
	440,000 LB Per 10,000 Landing LB Weight Above/Below 440,000 LB	Per 1,000 fl	Head/Tail wind	Down/Up Hill	Above/Below ISA	Per 5 KTS above VREF	One Reverser	No reverse	
Dry	5670	1104-90	170	-240/800	98/-80	150/-150	230	0	0
Wet (non- contaminated)	9300	170/-150	360	-480/1710	450/-350	290/-290	400	0	0
Wet(contaminated)	12400	280/-250	610	-790/2930	1010/-710	4804-470	540	0	0

Table 3:

100		I	Landing Dis	tance and A	djustments (f	feet)	12 3	0	
Runway condition	Reference Distance	Weight adjustment	Altitude adjustment ** Per 1,000	Wind adjustment per 10 knots Head/Tail wind	Slope Adjustment per 1% Down/Up Hill	Temperature Adjustment per 10° C Above/Helow ISA	Approach Speed Adjustment Per 5 KTS above VREF	Reverse Thrust Adjustment	
	420,000 LB Landing Weight	Per 10,000 LB Above/Below 420,000 LB						One Reverser	No reverse
Dry	5470	1004-90	160	-240/780	88/-80	150/-150	230	0	0
Wet (non- contaminated)	9010	170/-160	340	-470/1670	430/-340	290/-280	390	0	0
Wet(contaminated)	11740	270/-260	540	-750/2780	910/-650	460(-440)	530	0	

Table 4:

		1	anding Dist	tance and A	diustments (f	reet)			
	Reference Weight Altitude Wind Slope Temp Distance adjustment adjustment adjustment Adjustment Adjustment Adjustment per 10 per 1% per						Approach Speed Adjustment	Reverse Thrust Adjustment	
Runway condition 420,000 LI Landing Weight		Landing LB	Per 1,000 ft	Head/Tail wind	Down/Up Hill	Above/Below ISA	Per 5 KTS above VREF	One Reverser	No reverse
Dry	5500	1004-90	170	-240/790	98/-80	150(-150	230	0	. 0
Wet (non- contaminated)	9130	170/-150	360	-480/1700	450/-350	290/-280	390	0	0
Wet(contaminated)	11960	270/-260	590	-770/2860	940/-670	460(-460	530	0	0

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200		1	Landing Dist	tance and A	djustments (i	eet)		7	
	Reference Distance	adjustment adju	Altitude adjustment ++	Wind adjustment per 10 knots	Slope Adjustment per 1% Down/Up Hill	Temperature Adjustment per 10° C Above/Below ISA	Approach Speed Adjustment Per 5 KTS above VREF	Reverse Thrust Adjustment	
Runway condition	380,000 LB Landing Weight		Per 1,000 ft	Head/Tail wind				One Reverser	No reverse
Dry	5050	110/-80	150	-230/750	80/-70	130/-130	220	0	0
Wet (non- contaminated)	8370	170/-150	290	-440/1500	410/-320	250/-250	340	0	0
Wet(contaminated)	10800	290/-240	520	-720/2680	820/-590	430/-420	510	0	0

Table 6:

		1	anding Dist	tance and A	djustments (f	eet)			
	Reference Distance	adjustment adju	Altitude adjustment ++	Wind adjustment per 10 knots	adjustment Adjustment per 10 per 1%	Temperature Adjustment per 10° C Above/Below ISA	Approach Speed Adjustment Per 5 KTS above VREF	Reverse Thrust Adjustment	
	380,000 LB Landing Weight		Per 1,000 ft					One Reverser	No reverse
Dry	5100	110/-80	160	-230/760	80/-70	130/-140	220	0	0
Wet (non- contaminated)	8530	160/-140	300	-450/1530	430/-330	250/-250	340	0	0
Wet(contaminated)	11090	290/-240	560	-740/2790	880/-620	430/-420	510	0	0

Reference distance is based on MAX MANUAL braking, sea level, standard day, no wind or slope, and maximum reverse thrust.

Reference distance includes a distance from the threshold to touchdown associated with a flare time of 7 seconds.

Distances are based on HYD PRESS L+R failure distances which conservatively approximate the effects of 5G interference after the Reverse Thrust Adjustment for no Reversers is applied.

Actual (unfactored) distances are shown.

Note: per procedure, MAX MANUAL braking is not required for normal operations and is to be used only in the event that significant 5G interference occurs,

En route:

- Plan to use Flaps 30 and V_{REF30} (with appropriate wind additives) for landing.
- For wet runway conditions, compute the time of arrival (en route) landing distance
 using Tables 1 through 6, as applicable, applying all adjustments. Multiply the
 resulting unfactored landing distance by 1.1 to obtain the minimum required landing
 distance at the destination. This approximates a minimum required landing distance
 resulting from 5G C-Band interference.
- Determine the desired AUTOBRAKE setting by using the normal configuration landing distance information from an approved source, Maximum manual braking may not be required.

During Approach and Landing

- Monitor radio altimeters for anomalies.
- Normal use of autothrottles is allowed. Monitor the performance of autopilot and autothrottle. If the autopilot or autothrottle is not performing as expected, disconnect both the autopilot and autothrottle and apply manual inputs to ensure proper control of the flight path.
- If the autothrottle does not reduce the thrust to IDLE at 25 feet, manually reduce the
 thrust to idle, hold the thrust levers in the idle position, and disconnect the
 autothrottle to prevent the autothrottle from advancing the thrust levers after
 touchdown.

Caution: if the autothrottle advances the thrust levers after landing, the speedbrakes will stow and the autobrake will disarm. It will not be possible to raise the reverse thrust levers to deploy the thrust reversers until the thrust levers are at idle.

- Manual deployment of the speedbrakes may be required.
- If the thrust reversers do not deploy, immediately ensure the speedbrakes are extended, apply manual braking, and modulate as needed for the existing runway conditions.

Note: In some conditions, maximum manual braking may be needed throughout the entire landing roll.

NOTE 1: The AFM alteration required by this AD may be accomplished by inserting a copy of this AD into the Aircraft Flight Manual.

NOTE 2: For the purpose of this AD, a "radio altimeter tolerant aircraft" is the one for which ANAC accepts that the combination airplane-radio altimeter demonstrates tolerance to the limits specified in this PORTARIA No. 14.318/SAR, de 10 de abril de 2024, or further Portaria that supersedes it.

(3) For airplanes identified in paragraph (a) of this AD, that are defined as "radio altimeter tolerant aircraft", according to the established by PORTARIA No. 14.318/SAR, de 10 de abril de 2024, or further Portaria that supersedes it, no action is required by paragraph (b).

(c) Terminating Action to the AFM Revision

Modification of a "non-radio altimeter tolerant aircraft" to a "radio altimeter tolerant aircraft", according to PORTARIA No. 14.318/SAR, de 10 de abril de 2024, or further Portaria that supersedes it, terminates the AFM revision required by

paragraphs **(b)(1)** and **(b)(2)** of this AD. After modification to a "radio altimeter tolerant aircraft", according to PORTARIA No. 14.318/SAR, de 10 de abril de 2024, or further Portaria that supersedes it, remove the AFM revision required by paragraphs **(b)(1)** and **(b)(2)** of this AD.

(d) Compliance with PORTARIA No. 14.318/SAR, de 10 de abril de 2024, or further Portaria that supersedes it.

For the purpose of this AD, the acceptance of the combination airplane radio altimeter as a "radio altimeter tolerant aircraft" depends on evidence provided to demonstrate the tolerance limits established in PORTARIA No. 14.318/SAR, de 10 de abril de 2024, or further Portaria that supersedes it. These data should be submitted to ANAC through the e-mail 5g@anac.gov.br. The acceptance by ANAC of the airplane-radio altimeter combination as "radio altimeter tolerant aircraft" will be indicated through an ANAC Oficio to the operator or manufacturer, or an ANAC Portaria listing the configurations accepted by ANAC as "radio altimeter tolerant aircraft".

(e) Reporting of events

Report any anomalies in the radio altimeter to ANAC through the email 5g@anac.gov.br, providing the following information:

- (1) Date
- 2) Aircraft and radio altimeter model (3) Phase of flight (4) Location where the anomaly occurred (5) Transient or permanent anomaly (f) Alternative methods of compliance (AMOCs). A different method or a different compliance time from the requirements of this AD may be used if approved by the Manager of the Continuing Airworthiness Technical Branch (GTAC) of ANAC.

Record compliance with this AD in the applicable maintenance log book.

CONTACT:

For additional technical information, contact:

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APPROVAL:

ROBERTO JOSÉ SILVEIRA HONORATO Head of Airworthiness Department ANAC

NOTA: Original in Portuguese language signed and available in the files of the Continuing Airworthiness Technical Branch (GTAC) of the National Civil Aviation Agency (ANAC).

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